

M. B. ISMAILOV, I. K. ABLAKATOV, I. M. ALPYSBAY*

*National Center of Space Research and Technology, Almaty, Kazakhstan, *m.ismailov@spaceres.kz*

POSSIBILITIES of INTERMETALLIC COATING of Al-Cu and Cu-Zn OBTAINING by MAGNETRON SPUTTERING DEPOSITED at METAL SUBSTRATE

Abstract: The research aimed at investigating the possibility of obtaining thin intermetallic films, which can be used as thermal control coatings for spacecraft. The coating films build the passive part of the overall thermal control system of a spacecraft, thus they must provide strong adhesion with the material carrier and have high functional characteristics, optical ones in particular. The study concerned issues of synthesis of stable intermetallic phase of Al_4Cu_9 , Al_2Cu , Cu_5Zn_8 at aluminum and copper carriers by magnetron layer-wise sputtering of reagents. Regularities of intermetallic coating formation are studied in modes of "rapid" and "slow" sputtering of reagents, applying various thicknesses of sputtered layers of reagents, temperature of the carrier, heat treatment of sprayed coating. Incomplete and complete modes of synthesis of intermetallic coating have been discovered. Cross-sectional images of coatings, results of microanalyzer scanning of reagent distribution through the thickness of sputtered coating, microhardness values, optical absorption and emission ratios, unit of electric resistances, adhesion to the carrier were obtained. Prototypes of intermetallic thermostatic coatings in "solar reflectors" and "solar absorbers" classes were obtained. The results of measurements of optical and strength characteristics revealed that the intermetallic films can be used not only as thermoregulating coatings for space technology, but also in general mechanical engineering, due to high-end mechanical properties.

Keywords: sun, machine, thermoregulation, coating, synthesis, intermetallic compounds, magnetron sputtering, aluminum, copper, zinc.

Introduction. When in Earth orbit, a spacecraft (SC) is exposed to strong thermal radiation of the Sun leading to some individual surfaces heating up to $+150^\circ\text{C}$. When at the shadow side of the orbit, SC surfaces cool down to -150°C . Proper operation of SC electronics require comfortable temperatures close to the Earth ones. Required temperature is provided by and SC thermal regulation system, which protects electronics from overheating by solar radiation and heats colder parts of the working area. An important part of any thermal regulation system consists of thermostatic coatings (TSC) of a SC external surfaces [1]. TSCs are characterized by the following optical parameters: coefficient of absorption of solar radiation - A_s , degree of blackness or emissivity - ε [2].

The most important role belongs to the TSCs from "Solar reflectors" (also called "white") and "Solar absorbers" ("black") classes [2]. "Solar reflectors" are designed to combat overheating of an SC by solar radiation.

Body temperature under the influence of direct solar radiation can be calculated by using the following connection [3]:

$$T = (J_s A_s / \sigma \varepsilon)^{1/4} \quad (1)$$

where T - equilibrium heating temperature, J_s - solar radiation heat flux, σ - Stefan-Boltzmann constant.

Dependence (1) demonstrates that minimizing temperature of SC heating by solar radiation is achieved by minimization of coefficient ratios of A_s/ε TSC, meaning the less A_s and the more ε is, the less the body temperature becomes. TSC criterion for "Solar reflectors" class is the ratio of [4]:

$$A_s/\varepsilon \leq 1 \quad (2)$$

TSC class "Solar absorbers" is designed for maximum absorption of solar energy, which further is transported inside the SC to provide thermal energy to individual units. Such TSC require large values of A_s/ε .

TSC requirements are as follows: required optical characteristics of A_s and ε , resistance to oxidation by oxygen ions and space plasma, resistance to ionizing radiation of the Sun, electrical conductivity for removal of static electricity, strength (hardness and adhesion to the carrier) to counteract space particles [3].

"White" paint for TSC are produced from oxide powders and resins, they get sputtered upon the surface of the SC parts. «White» TSCs come also in the form of organic films or glass sputtered from the backside of the metal. "Black" paint for TSCs are made from black pigments and resins.

It is known that in the space outer space factors (OSF) have a strong degrading impact on TSC leading to significant changes in the A_s coefficient, in this case, as it turned out, the ε coefficient remains virtually constant.

Table 1 shows examples of “white” industrial TRP, as well as their A_s values in the conditions of outer space. As the data shows (experiments over 5 years, whilst SC fly for 15 years), TSC degradation especially actively happens under geostationary orbit conditions, it leads to gradual heating of SC [1].

Experts believe that new TSC classes, particularly class “Solar reflectors” with stable thermal radioactive properties at long-term service when applied to SC in space, are one of the most important priorities in the space industry of the 21 century [4]. Creation of such TSC will increase lifetime of an SC by more than 15 years.

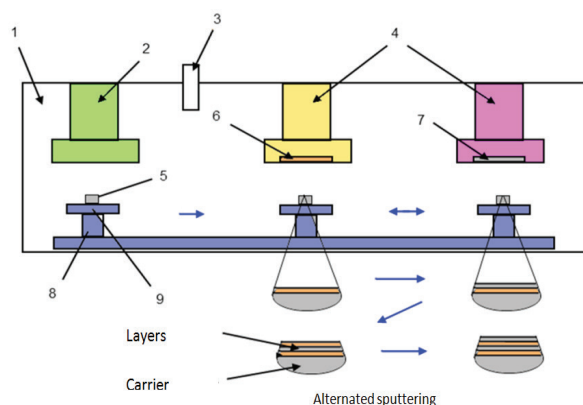
In order to achieve that there have been being conducted continuous search of innovative TSC types. Intermetallic compounds are of outlook interest in regards with TSC, they are distinguished by the variety of colors, durability, and corrosion resistance to oxidation by oxygen, and conductivity [5-7]. Literature and patent study showed a lack of research in the area of applying intermetallic compounds as TSC.

Table 1 - Experimental data on degrading of “white” TSC at SC under outerspace conditions (L - low orbits, GSO-geostationary orbit)

| TSC type | | Composition, thickness, μm | Temperature range, $^{\circ}\text{C}$ | Initial coefficients | | As condition during the operation of an SC in orbits | | | |
|-----------|-------------|---|---------------------------------------|----------------------|------------|--|---------|----------------|---------|
| | | | | As | ϵ | L < 1 000 km | | GSO ~36 000 km | |
| | | | | | | 3 years | 5 years | 3 years | 5 years |
| LCP | AK-512 | White enamel, 100 μm | -150 +150 | 0.3 | 0.85 | 0.32 | 0.36 | 0.62 | 0.67 |
| | AK-573 | Same | -150 +200 | 0.24 | 0.85 | 0.21 | 0.21 | 0.45 | 0.5 |
| Cera-mic | TS-SO-1 | Aqueous solution of potassium silicate, 240 μm | -100 +150 | 0.19 | 0.92 | 0.16 | 0.17 | 0.45 | 0.5 |
| | TP-SO-10 | | -100 +150 | 0.18 | 0.9 | 0.09 | 0.09 | 0.57 | 0.6 |
| Thin film | OSO-A | Glass plate K-208 with metallic 200 μm | -100 +100 | 0.13 | 0.9 | 0.17 | 0.18 | 0.26 | 0.28 |
| | OSO-S | | -100 +100 | 0.13 | 0.85 | 0.17 | 0.18 | 0.31 | 0.32 |
| | SOT 1-A-100 | Film F-4MB | -100 +100 | 0.17 | 0.8 | 0.13 | 0.13 | 0.37 | 0.42 |
| | SOT-1-100 | | -100 +100 | 0.11 | 0.85 | 0.13 | 0.13 | 0.29 | 0.35 |

This paper topic was defined for this same reason; it is devoted to obtaining intermetallic coatings and study of their characteristics as TSC test pieces. The coatings were created by magnetron sputtering on a copper or aluminum carrier of alternating thin (nanoscale) layers of pairs of metal reagents Al-Cu and Cu-Zn. The goal is to obtain coatings made of sustainable intermetallic Al_4Cu_9 , Al_2Cu , Cu_5Zn_8 per the diffusion reaction mechanism. This technique is known as the method of coating flexible surfaces with strengthening intermetallic coatings [8], however, the possibility of using such coatings as TSC has not been investigated.

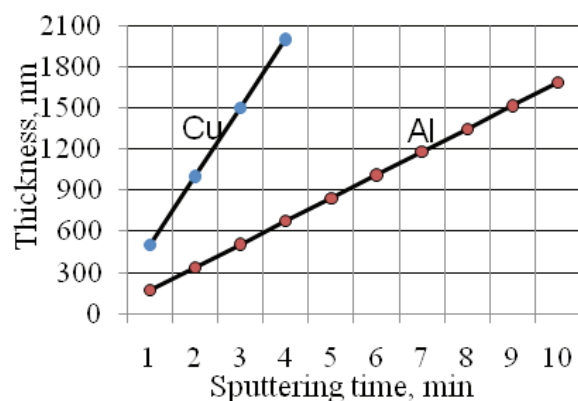
Experimental Part. Intermetallic coatings were generated using magnetron unit owned by the “Center for Earth Sciences, metallurgy and refining” JSC (CESMR) the unit schema is shown in Figure 1. Aluminum with 99.995 % purity, copper with 99.997 % purity and zinc with 9.995 % purity were selected as targets for sputtering. Target diameter was 110 mm, its thickness was 5 mm.



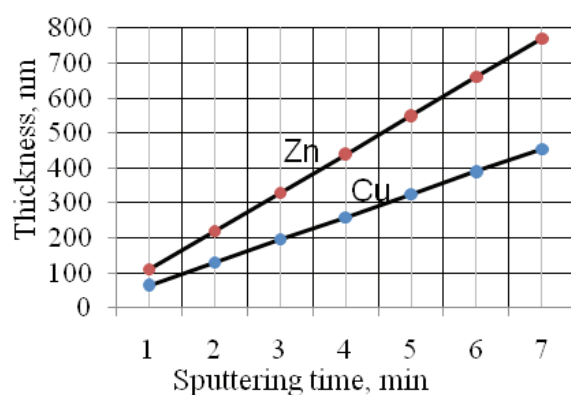
1 - vacuum chamber, 2 - ion cleaning system, 3 - argon system, 4 - magnetrons, 5 - carrier, 6 and 7 - targets for sputtering, pumping system, 8 - sliding system, 9 - carrier holder

Figure 1 - Diagram of magnetron unit

There have been implemented 2 metal spraying regimes: rapid and slow with thinner layers of reagents. Current-voltage characteristics of magnetron for these modes are presented in Table 2. To calibrate the magnetron one common admission about the sameness of thicknesses of magnetron etal sputtering on metal and glass carrier have been considered. Glass carrier gives the opportunity to measure the thickness of sprayed layer using atomic force microscope JSPM-5200, sputtering time was measured with a chronometer. Thickness calibration results for Al and Cu, Zn for rapid and slow coating modes are show in Figures 2, 3.



a – Thickness of Cu-Al



b – Thickness of Cu-Zn

Figure 2 - Thickness calibration depending on sputtering time for one layer at rapid coating mode

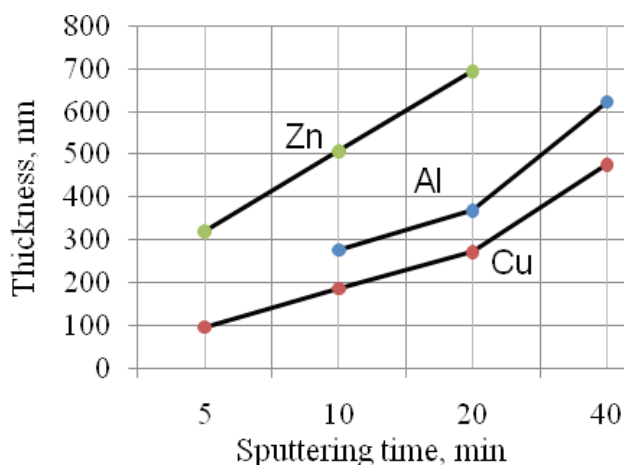


Figure 3 - Thickness calibration depending on sputtering time for one layer at slow coating mode

Sputtering preparation process was carried out by vacuuming the operating chamber, cleaning the surface of the target and the carrier by ionic treatment, and setting voltage and current

characteristics. Metal carrier coating of required thickness was performed according to the calibration data in Figure 3, according to the sputtering time. It was expected to achieve the necessary phase by adjusting the thicknesses (mass) of reagent layers. The objective was to obtain sustainable intermetallic phases Al_4Cu_9 and Al_2Cu in Al-Cu coatings system of phase Cu_5Zn_8 to coat the Cu-Zn system. The objective should be achieved by sputtering stoichiometric ratio of the reagent masses, obtainable by choosing the thickness ratio of coating layers. Reagent thickness ration was calculated per the following connections: for phase Al_4Cu_9 - $h_{Cu} = 1,58h_{Al}$; for Al_2Cu - $h_{Cu} = 0,353h_{Al}$; for Cu_5Zn_8 - $h_{Cu} = 0,482h_{Zn}$. Experimenting plan is given in Table 2.

Table 2 - Experimenting plan

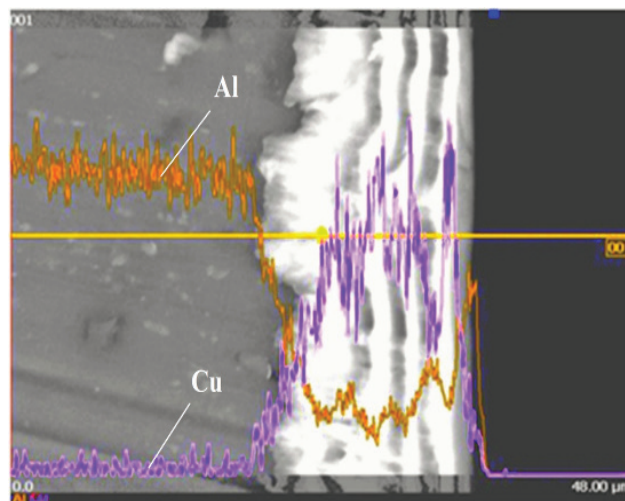
| Coating № | Magnetron setting mode | Current-voltage characteristics of the magnetron | Synthesized phases | Carrier temperature, °C | Coating layer thickness, nm | Number of layer pairs |
|-----------|------------------------|--|--------------------|-------------------------|-----------------------------|-----------------------|
| 1 | Rapid Sputtering | Al - 360 V, 1A Cu - 400 V, 2A | Al_4Cu_9 | 25 | Al – 1200 Cu – 1900 | 4 |
| 2 | | | Al_4Cu_9 | 25 | Al – 600 Cu – 950 | 6 |
| 3 | | Zn-480V, 0.27A CU-371V, 1A | Cu_5Zn_8 | 25 | Cu – 260 Zn – 550 | 4 |
| 4 | | | Cu_5Zn_8 | 25 | Cu – 260 Zn – 550 | 6 |
| 5 | Slow Sputtering | Cu – 330 V, 0.2A Al – 300 V, 0.12A | Al_2Cu | 200 | Al – 276 Cu – 96 | 2 |
| 6 | | | Al_2Cu | 200 | Al – 368 Cu – 185 | 2 |
| 7 | | | Al_2Cu | 200 | Al – 610 Cu – 220 | 2 |
| 8 | | | Al_2Cu | 200 | Al – 610 Cu – 220 | 4 |
| 9 | | Zn – 300V, 0.1A Cu – 330 V, 0.2A | Cu_5Zn_8 | 200 | Cu – 185 Zn – 320 | 2 |
| 10 | | | Cu_5Zn_8 | 200 | Cu – 217 Zn – 507 | 2 |
| 11 | | | Cu_5Zn_8 | 200 | Cu – 475 Zn – 694 | 2 |
| 12 | | | Cu_5Zn_8 | 200 | Cu – 217 Zn – 507 | 4 |

The resulting samples were tested to identify the following characteristics of the coatings: cross-sectional images and distribution of concentrations of reagents coated in terms of thickness (using micro analyzer of an electron-scanning probe JXA-8230 [9]), phase composition by frontal surface x-ray diffractometer D8 ADVANCE [10], values of microhardness using microhardness-meter PMT-3 [11] (by indentation of a diamond pyramid indenter under a load of 5 g into the coating surface), light absorption ratios A_s for wavelength range of 250-2400 nm using spectrophotometer Shimadzu UV-3600 [12], emissivity ε using infrared pyrometer UNIT UT-302B [13] for wavelength range of 800-1400 nm. Coating adhesion to the carrier was evaluated by scratching (GOST 9.302-88 "Metallic and non-metallic mineral coatings. Methods of control"), electric tester was used to evaluate conductivity.

The coating obtained by rapid sputtering were studied at room temperature, as well as after treating them with heat and subsequent cooling: for the Al-Cu system 450 °C for one hour, for the Cu-Zn system 350 °C for one hour. The coating obtained by slow sputtering were tested at room temperature, and some extra heat treatment was applied in some cases, at a temperature of 150 °C for one hour.

Results and Discussion. 1. *Coatings obtained by rapid sputtering.*

1.1 Al-Cu System. Experiments were conducted using an aluminum carrier. Figure 4 shows a snapshot of the cross-section of coating № 1 (Al-Cu System), as well as the distribution pattern of concentration of Al and Cu reagents across the coating thickness. Snapshot reveals strips of reagents, as well as the blurred boundary between the carrier and the coating (the result of spreading reagents during polishing the sample, as well as mutual diffusion of the coatings and the carrier), average thickness is estimated at 12.2 μm that is close enough to the calculated thickness of 12.4 μm . Readings of Al and Cu concentrations across the thickness indicate that in reality the coating consists of Al and Cu layers, however concentrations of these elements in the layers is slightly lower than the one achieved in sputtering mode (comparison of the Al concentration in the carrier and the coating), which means partial reaction of intermetallic formation. The results indicate that sputtering reagent layers onto the carrier keeps their individuality, the reaction of intermetallic formation was not completed. To complete the synthesis of intermetallides in the coating, it requires to be treated with heat.



Light section - the coating, section to the left - the carrier. Coating thickness is 12.2 μm .

Figure 4 - Cross-sectional snapshot of coating № 1

Cover № 2 displayed results similar to the results obtained for coating № 1. To trigger the reaction of intermetallic formation the coating № 2 was subjected to heat treatment in vacuum at a temperature of 450°C for one hour. Snapshot of cross-section of coating № 2 after heat treatment is shown in Figure 5.

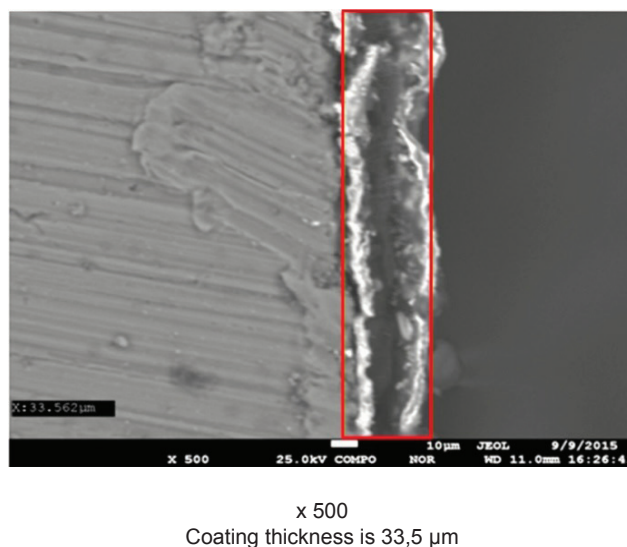


Figure 5 - Snapshot of cross-section of coating № 2 after heat treatment

It can be seen that the sandwich structure of the coating is gone, the coating surface and the border with the carrier have light stripes, presumably the stripes indicate presence of unreacted Cu. After thermal treatment the coating thickness has tripled, the reason for this phenomenon is not clear. Table 3 shows the results of XRD of coating № 2.

Table 3 - XRD Data for coating №2 with aluminum carrier

| Phase | Formula | Content, % | |
|-------------------------------------|---------------------------------|-------------------------|----------------------|
| | | prior to heat treatment | after heat treatment |
| Aluminum | Al | 69.2 | 81.8 |
| Aluminum in copper (solid solution) | Al-5.6%, Cu-94.3 | 22.4 | 5.4 |
| Copper in aluminum (solid solution) | Al-99%, Cu-1% | 8.4 | 7.2 |
| Intermetallides | Al ₄ Cu ₉ | - | 5.7 |

For interpretation of the data in the Table 3 it must be noted that the frontal radiography of thin coating penetrates the aluminum carrier. In view of this, the “Aluminum” phase refers to the carrier, the other phases can be attributed to the coating.

It can be seen that before the heat treatment, of the coating, the Al₄Cu₉ phase is practically not visible, it appears after the heat treatment. Incompleteness of the intermetallic formation reactions in the coating is also obvious due to remnants of Al and Cu in the form of solid solutions. The most likely cause of incomplete synthesis is the excessive thickness of the reagent layers. Coating hardness after heat treatment is low - about 40 MPa.

1.2 Cu-Zn System. The results of XRD of coating №3 are shown in Table 4.

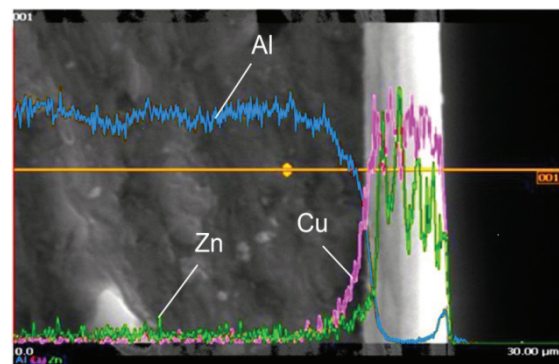
Table 4 - The results of XRD of coating №3

| Phase | Formula | Content, % | |
|-------------------------------------|---|-------------------------|----------------------|
| | | prior to heat treatment | after heat treatment |
| Aluminum | Al | 71.6 | 64.8 |
| Intermetallides | Cu ₅ Zn ₈ | 11.4 | - |
| Aluminum in copper (solid solution) | Al-5.6%, Cu-94.3% | 9.9 | - |
| Copper in aluminum (solid solution) | Al-99%, Cu-1% | 7.1 | - |
| Intermetallides | Al _{0.0565} Cu _{0.9434} | - | 16 |
| Intermetallides | Cu _{1.05} Zn _{0.95} | - | 10.6 |
| Intermetallides | (Al _{0.99} Cu _{0.01}) | - | 5.7 |
| Intermetallides | Cu _{0.70} Zn ₂ | - | 1.6 |
| Intermetallides | Cu _{0.7} Zn _{0.3} | - | 1.3 |

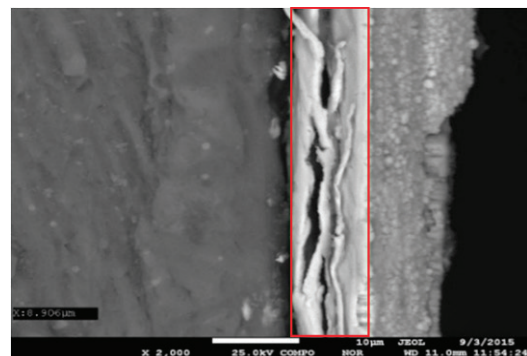
“Aluminum” phase in table 4 should be attributed to the carrier, the remaining phases are related to the coating. It is obvious that a significant number of intermetallic Cu₅Zn₈ is formed during sputtering the coating; however, the process remains incomplete. Heat treatment of the coating at 350°C for one hour led to formation of a number of unintended and fragile intermetallides, including aluminum from the carrier.

During the heat treatment, the coating increased considerably in thickness.

Figure 6 shows cross-section snapshot of coating № 4. Here as well there are the results of the micro analyzer scan of presence of Zn, Cu reagents as well as Al from the carrier. There are clearly visible layers of reagents, as well as the mutual diffusion penetration of carrier and coating elements. After thermal treatment of the sample in vacuum at 350°C for one hour (figure 6b), the coating thickened from the original value of 4.8 μm to 8.9 μm, along with formation of cracks and voids. Thicker coatings during thermal processing was caused by a complex process of chemical and physical interaction of intermetallic Cu₅Zn₈, as well as residual Cu and Zn, with aluminum carrier.



a - prior to heat treatment thickness is 4,8 μm.



b - after heat treatment, thickness is 8,9 μm.

Figure 6 - Cross-sectional images of the coating № 4

Thus, during sputtering coatings № 3 and 4 the synthesis of intermetallic phase Cu₅Zn₈ happens only partially, considerable quantities of reagents remain in the form of solid solutions. Completion of intermetallic formation is hindered by thick layers of reagents. When heat-treated, under the influence of active aluminum from the carrier, primary sputtered products transform into a number of unstable intermetallides, the coating thickens and cracks and voids occur. Hardness of the coatings is

low - 41 MPa.

2. Coatings obtained by slow sputtering

Experiments were conducted on a copper carrier according to experimental plan shown in Table 2. Sample coatings №5-12 were manufactured. Coatings from this series had smaller thickness than the ones in rapid sputtering mode, which complicated research of the distribution of reagents across the coating thickness, as well as getting the XRD. The real method of characterizing the coating is assessing microhardness. Table 5 shows test results.

Experimental results have shown:

- coatings № 5-7, 9-11, which were obtained by combining 2 pairs of layers and were less than 2.9 μm thick, have the same microhardness after sputtering as the original metal microhardness, and after thermostating at 150 $^{\circ}\text{C}$ for one hour, their microhardness increases by an order. These data indicate that intermetallides do not form during sputtering; they are synthesized during heat treatment;

- coatings № 8 and 12, obtained from 4 pairs of reagents and having thickness equal to 2.9 μm or more, have a very high microhardness of 6000 MPa immediately after sputtering, heat treatment does not change this value. The resulting value of microhardness is normal for intermetallides [14], hence, it can be concluded that intermetallic compounds form during sputtering.

Table 5 - Strength characteristics of coatings obtained in slow sputtering mode

| Coating № | Anticipated phase composition | Number of sputtered layer pairs | Coating thickness, μm . | Microhardness, MPa | |
|-----------|-------------------------------|---------------------------------|------------------------------------|-----------------------|----------------------|
| | | | | before heat treatment | after heat treatment |
| 5 | Al_2Cu | 2 | 0.74 | 204 | 2430 |
| 6 | Al_2Cu | 2 | 1.1 | 220 | 2570 |
| 7 | Al_2Cu | 2 | 1.66 | 264 | 2600 |
| 8 | Al_2Cu | 4 | 3.32 | 6000 | 6000 |
| 9 | Cu_5Zn_8 | 2 | 1.01 | 85 | 970 |
| 10 | Cu_5Zn_8 | 2 | 1.45 | 170 | 1210 |
| 11 | Cu_5Zn_8 | 2 | 2.34 | 185 | 1340 |
| 12 | Cu_5Zn_8 | 4 | 2.90 | 2600 | 2400 |

These data suggest that there is a certain critical thickness of coating sputtered to carrier. If thickness value is less than this critical thickness, then intermetallic compounds form poorly. The most likely cause of this phenomenon is the effect of carrier, which diffuses with the sputtered layer and forms a solid solution with the latter, thereby passivates chemical activity of the reagent. Reaction triggering requires heat treatment. Increase in coating thickness leads to carrier influence weakening and acceleration of the intermetallide formation reaction.

To generalize the results, the following conditions

of synthesizing intermetalline coatings during magnetron sputtering of reagents can be derived:

- reagent sputtering should be slow with the magnetron setting according to the order quoted in Table 2 and Figure 3;

- carrier must be heated to the temperature of approximately 200 $^{\circ}\text{C}$,

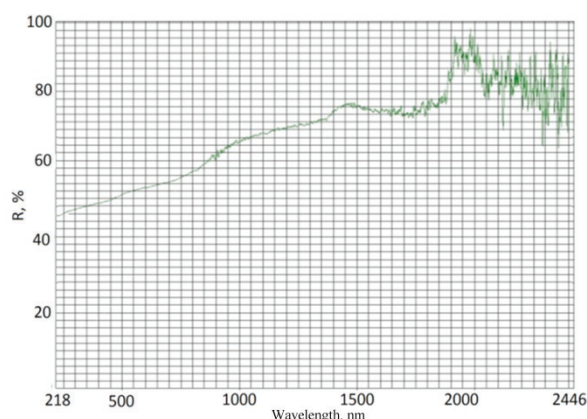
- thickness of sputtered reagent layer should be fairly thin - it should not exceed 610 nm,

- the number of pairs of sputtered reagents must be at least 4, with a total thickness of coating more than 2.9 μm .

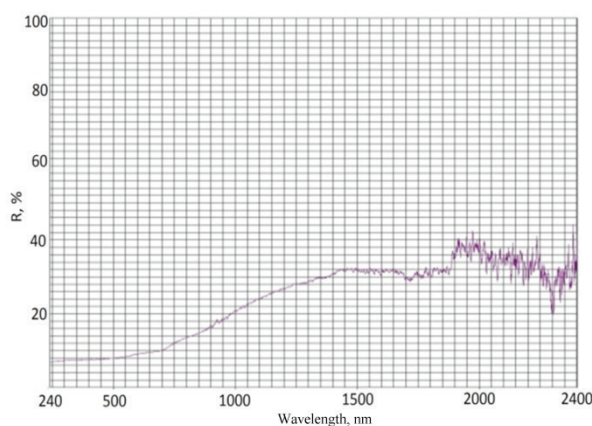
Strictly speaking, these conditions only apply to the reviewed coatings systems; nonetheless they can be used as benchmarks for other systems.

3. Optical properties of coatings

There have been defined optical coefficients A_s and ϵ for coatings №2 and №4, which were obtained in the rapid sputtering mode, as well as for coatings № 8 and №12 from the slow mode. Figure 7 shows examples of experimental data of reflection coefficient R (%) depending on the lengths of optical waves for coatings № 8 and 12.



a – coating № 8



b – coating № 12

Figure 7 - Reflection coefficient of the coatings by wavelength

Average integral A_s ratio is defined in the wavelength range of 240-2400 nm. As Calculation was based on the ratio $A_s = 1-R/100$. Experimental results are summarized in Table 6. Unit resistivity of individual phases was taken from cited sources [15-17], followed by validation using electric resistance coating tester, which confirmed the order of these data. All coatings have sufficient electrical conductivity to dissipate any static electricity. The coatings are diffused and spliced with the carrier, which is well displayed by figures 5 and 6. Scratching the coating does not lead to its scaling from the carrier.

Table 6 - Optical, electrical and adhesive characteristics of coatings

| Coating № | Sputtering mode | Phase composition | Micro-hardness, MPa | As | ε | As/ ε | Resistivity, micro Ohm/cm | Adhesion to carrier |
|-----------|-----------------|-------------------|---------------------|------|---------------|-------------------|---------------------------|---------------------|
| 2 | Rapid | Al_4Cu_9 | 43 | 0.82 | 0.44 | 1.96 | 14.2 [15] | Excellent |
| 4 | | Cu_5Zn_8 | 42 | 0.79 | 0.38 | 2.1 | 11.1 [17] | Excellent |
| 8 | Slow | Al_2Cu | 6000 | 0.3 | 0.49 | 0.61 | 7.0 [15] | Excellent |
| 12 | | Cu_5Zn_8 | 2600 | 0.79 | 0.38 | 2.1 | 11.1 [17] | Excellent |
| | | Al | 263 | | | | 2.4 [15] | |
| | | Cu | 818 | | | | 2.0 [15] | |
| | | Zn | 570 | | | | 5.9 [16] | |

The figures in table 6 indicated that the following coatings can be taken as TSC prototypes:

- coating № 8 in the “Solar reflectors” category (“white”),
- coating № 12 in the “Solar absorbers” category (“black”),

Indeed, coating № 8 meets the requirement (2), has very high hardness, sufficient electrical conductivity. Intermetallide phases are corrosion resistant [14, 15, 18, 19]. Coating №12 absorbs energy from the Sun well, radiates it moderately, is durable and has sufficient electrical conductivity. Further research could focus on improving characteristics of the coatings.

Findings of this research shall be of particular interest to separate task of creating strengthening intermetalline coatings in mechanical engineering.

The authors are grateful to the staff of the “Center for Earth Sciences, metallurgy and refining» JSC, Mr. A.V. Panichkin and Ms. A.A Mamayeva, for facilitating this work.

Conclusions. The aim is the synthesis of thermostatic coating prototypes for SC, which are to be based on metal carrier and consist of sustainable intermetalline phases Al_4Cu_9 , Al_2Cu , Cu_5Zn_8 . In order to synthesize it was decided to use the method of reaction diffusion among alternating layers of two reagents, which are to be sputtered upon a metal carrier by a magnetron and have to have thickness values compliant to stoichiometry of the phase being synthesized.

There have been used 2 reagent sputtering modes: “rapid” with the carrier temperature being equal to room temperature, and “slow” with a carrier heated to 200 °C. Configuring the magnetron and production of experimental samples of coatings was conducted according to experimental plan set out in Table 2.

Modes of incomplete and complete synthesis of intermetallic coating were identified. Authors formulated conditions for intermetallic synthesis during reagent sputtering.

Acceptable results in terms of coating quality were obtained during the mode of slow sputtering. At the same time:

- coating № 8, which is 3.3 μm thick, with phase compound of Al_2Cu , has high microhardness of 6000 MPa, optical absorption coefficient $As = 0.3$, emissivity $\varepsilon = 0.49$, color criterion $As/\varepsilon = 0.61$, electrical resistivity of 7 microOhm/cm, excellent adhesion to the carrier, according to the sources - high chemical resistance. This coating can be used as the prototype of the “Solar reflectors” TSC class;

- coating № 12, which is 2.9 μm thick, with phase compound of Cu_5Zn_8 , has high microhardness of 2600 MPa, optical absorption coefficient $As = 0.79$, emissivity $\varepsilon = 0.38$, color criterion $As/\varepsilon = 2.1$, electrical resistivity of 2.1 microOhm/cm, excellent adhesion to the carrier, according to the sources - high chemical resistance. This coating can be used as the prototype of the “Solar absorbers” TSC class.

The work is based on grant project funded by MES of RK № 0094/TΦ4.

REFERENCES

- 1 Korolev S.Y. *Sistemy obespecheniya teplovogo rezhima kosmicheskikh apparatov*. (Systems of thermal controlling of spacecrafts). *Baltiyskij universitet «Voenmekh»* (University of Baltic «Voenmekh»). Moscow: Mir, **2006**. 100 (in Russ.).
- 2 Mikhailov M.M. *Prognozirovanie opticheskoi degradatsii termoreguliruyushchikh pokrytij kosmicheskikh apparatov* (Prediction of optical degradation of thermal control coatings of spacecraft). Novosibirsk: Nauka, **1999**. 192 (in Russ.).
- 3 Novikov L.S. *Model' kosmosa. Vozdejstvie kosmicheskoi sredy na materialy i oborudovaniya kosmicheskikh apparatov* (The Space model. Impact of the space environment on materials and spacecraft equipment). Moscow: KDU, **2007**. 2. 1144 (in Russ.).

4 Khalimanovich V.I., Kharlamov V.A., Ermolaev R.A. *Ispytaniya laboratornykh obraztsov termoreguliruyushchikh pokrytij ugleplastikovykh ehlementov kosmicheskikh apparatov* (Testing of laboratory sample of thermal control coatings of carbon elements of spacecraft) [Electron resource]. – URL: <http://elibrari.ru/item.asp?id=13611495> (date of access: 21.06.2016) (in Russ.).

5 Grinberg B.A. *Intermetallidy: fundamental'nye aspekty, prilozheniya* (Intermetallic compounds: fundamental aspects, applications) [Electron resource]. – URL: <http://www.uran.ru/reports/t80.htm>. (date of access: 21.06.2016) (in Russ.).

6 *Dominant Intermetallic Compounds for Common Bimetal Combination* [Electron resource]. – URL: <http://web.cecs.pdx.edu>. (date of access: 21.06.2016) (in Eng.).

7 Yanhong Tian, Chunjin Hang, Chuqing Wang, Y.Zhou. Evolution of Cu/Al Intermetallic Compounds in the Copper Bump bonds during Aging Process [Electron resource]. – URL: <http://ieeexplore.ieee.org>. (date of access: 21.06.2016) (in Eng.).

8 Kushchev C.B., Maksimenko A.A., Bosykh M.A. *Tverdosť plenok sistemy Al-Cu* (Hardness of films of Al-Cu systems). *Kondensirovanie sredy i mezhfaznykh granits = Condensed matter and interphase boundaries* 1998. 14. 1. 53-59 (in Russ.).

9 *Ehlektronno-zondovyy rastrovyy mikroskop JXA-8230 s mikroanalizatorom* (Electron probe scanning microscope JXA-8230 with Microanalyzer) [Electron resource]. – URL: <http://www.eavangard-semi.ru/jeoljxa8230> (date of access: 21.06.2016) (in Russ.).

10 *Rentgenovskiy Difraktoметр D8 Advanced* (X-Ray Diffraction D8 Advanced) [Electron resource]. – URL: http://www.ufaras.ru/part_id=398,401,417 (date of access: 21.06.2016) (in Russ.).

11 *Microtverdomer PMT-3* (Microhardness PMT-3) [Electron resource]. – URL: <http://www.asma-pribor.ru/dmodules/downloads/download/12/54/> (date of access: 21.06.2016) (in Russ.).

12 *Spectrofotometr Shimadzu UV-3600 Plus* (Spectrophotometer Shimadzu UV-3600 Plus) [Electron resource]. – URL: <https://www.shimadzu.ru/uv-3600-plus> (date of access: 21.06.2016) (in Russ.).

13 *Pirometr UNIT UT-302B* (Pyrometer UNIT UT-302B) [Electron resource]. – URL: <http://f77.kz/p3540506-beskontaktnyj-termometr-pirometr.html> (date of access: 21.06.2016) (in Russ.).

14 Adeline B.Y.Lim, Xin Long, Lu Shen. Effect of Palladium on the Mechanical Properties of Cu-Al Intermetallic Compounds. *Journal of Alloys and Compounds*. 2015. 628. 107-112 (in Eng.).

15 Drozdov Maria. Microstructural Evolution of Al-Cu Intermetallic Phases in Wire-Bonding: *thesis for Master's Degree in Materials Engineering Science*. Technion – Ins. of Technology. Haifa. Israel. 2007. 94. (in Eng.).

16 *Udel'noe Ehlektricheskoe soprotivlenie* (Electrical Resistivity) [Electron resource]. – URL: https://ru.wikipedia.org/wiki/Удельное_электрическое_сопротивление (date of access: 21.06.2016) (in Russ.).

17 Mansoor Farbod, Alireza Mohammadian. Effect of Sintering on the Properties of γ -Brass (Cu₅Zn₈) Nanoparticles Produced by the Electric Arc Discharge Method and the Thermal Conductivity of γ -Brass Oil-Based Nanofluid. *Metallurgical and Materials transactions A*. 2016. 47a. 1409-1412 (in Eng.).

18 Lyacine Aloui, Thomas Duguet, Fanta Haidara. Al-Cu intermetallic coatings processed by sequential metalorganic chemical vapour deposition and post-deposition annealing. *Applied Surface Science*. 2012. 258. 6425-6430 (in Eng.).

19 Kwang Seok Lee, Yong-Nam Kwon. Solid-state bonding between Al and Cu by vacuum hot pressing.

Transactions of Nonferrous Metals Society of China. 2013. 23. 341-346 (in Eng.).

ЛИТЕРАТУРА

1. Королев С.И.. Системы обеспечения теплового режима космических аппаратов. Балтийский университет «Военмех». – М.: Мир, 2006. – 100 с.

2. Михайлов М.М. Прогнозирование оптической деградации терморегулирующих покрытий космических аппаратов. – Новосибирск: Наука, 1999. – 192 с.

3. Новиков Л.С. Модель космоса. Воздействие космической среды на материалы и оборудования космических аппаратов. – М.: КДУ, 2007. – Т. 2. – 1144 с.

4. Халиманович В.И., Харламов В.А., Ермолаев Р.А. Испытания лабораторных образцов терморегулирующих покрытий углепластиковых элементов космических аппаратов [Электронный ресурс]. – URL: <http://elibrari.ru/item.asp?id=13611495> (дата обращения: 21.06.2016)

5. Гринберг Б.А. Интерметаллиды: фундаментальные аспекты, приложения [Электронный ресурс]. – URL: <http://www.uran.ru/reports/t80.htm>. (дата обращения 21.06.2016).

6. Dominant Intermetallic Compounds for Common Bimetal Combination [Электронный ресурс]. – URL: <http://web.cecs.pdx.edu>. (дата обращения 21.06.2016).

7. Yanhong Tian, Chunjin Hang, Chuqing Wang, Y.Zhou. Evolution of Cu/Al Intermetallic Compounds in the Copper Bump bonds during Aging Process [Электронный ресурс]. – URL: <http://ieeexplore.ieee.org>. (дата обращения 21.06.2016).

8. Куцев С.Б., Максименко А.А., Босых М.А. Твердость пленок системы Al-Cu. // Конденсирование среды и межфазных границ – 1998. – Т. 14. №1. – С. 53-59.

9. Электронно-зондовый растровый микроскоп с микроанализатором JXA-8230 [Электронный ресурс]. – URL: <http://www.eavangard-semi.ru/jeoljxa8230> (дата обращения 21.06.2016).

10. Рентгеновский Дифрактометр D8 Advanced [Электронный ресурс]. – URL: http://www.ufaras.ru/part_id=398,401,417 (дата обращения 21.06.2016).

11. Микротвердомер ПМТ-3. [Электронный ресурс]. – URL: <http://www.asma-pribor.ru/dmodules/downloads/download/12/54/> (дата обращения 21.06.2016).

Спектрофотометр Shimadzu UV-3600 Plus [Электронный ресурс]. – URL: <https://www.shimadzu.ru/uv-3600-plus> (дата обращения 21.06.2016).

Пирометр UNIT UT-302B [Электронный ресурс]. – URL: <http://f77.kz/p3540506-beskontaktnyj-termometr-pirometr.html> (дата обращения 21.06.2016).

12. Adeline B.Y.Lim, Xin Long, Lu Shen. Effect of Palladium on the Mechanical Properties of Cu-Al Intermetallic Compounds // *Journal of Alloys and Compounds* 628. – 2015. – P. 107-112

13. Drozdov Maria. Microstructural Evolution of Al-Cu Intermetallic Phases in Wire-Bonding: *thesis for Master's Degree in Materials Engineering Science*. / Technion – Ins. of Technology. – Haifa, Israel, 2007. – 94 p.

14. Удельное электрическое сопротивление [Электронный ресурс]. – URL: https://ru.wikipedia.org/wiki/Удельное_электрическое_сопротивление (дата обращения 21.06.2016).

15. Mansoor Farbod, Alireza Mohammadian. Effect of Sintering on the Properties of γ -Brass (Cu₅Zn₈) Nanoparticles Produced by the Electric Arc Discharge Method and the Thermal Conductivity of γ -Brass Oil-Based Nanofluid //

Metallurgical and Materials transactions A. -2016. – V. 47a. – P. 1409-1412.

16. Lyacine Aloui, Thomas Duguet, Fanta Haidara. Al-Cu intermetallic coatings processed by sequential metalorganic chemical vapour deposition and post-

deposition annealing // Applied Surface Science, - 2012. - V. 258. - P. 6425-6430.

17. Kwang Seok Lee, Yong-Nam Kwon. Solid-state bonding between Al and Cu by vacuum hot pressing // Transactions of Nonferrous Metals Society of China. - 2013. - V. 23. - P 341-346.

ТҮЙІНДЕМЕ

Ғарыш аппараттарында қолданылатын интерметаллитті жылуреттеуіш жабындыларын алу жолдары зерттелді. Реагенттерді келме-кезек магнетрон шашырату арқылы алюминий мен мыстан жасалған төсемге қоңдырылған төзімді Al_4Cu_9 , Al_2Cu , Cu_5Zn_8 интерметаллиттік фазаларынан тұратын жабындыларын синтездеу сұрақтары қарастырылған. Жабындыны термиялық өңдеу, төсемнің температурасы, реагенттерді шашырату арқылы алынған түрлі қылындықтағы қаптамалар, реагенттерді жылдам және баяу шашырату арқылы интерметаллиттік жабындыларын түзу заңдылықтары зерттелген. Интерметаллиттік жабындыларының толық және жартылай синтезделу процесстері байқалған. Жабындының микронзондық сканирлеу көлденең қимасының суреттері, микроқаттылық өлшемдері, сәулелену және жұту оптикалық коэффициенттері, меншікті электр кедергілері мен төсемге жабысулық нәтижелері көрсетілген. «Күн сәулесін шағылдырғыш» және «күн сәлесін жұтатын» интерметаллиттік жылуреттеуіш жабындыларының түпнұсқалары алынды.

Түйінді сөздер: күн, аппарат, термореттеуіш, жабынды, интерметаллиттер, магнетрон, шашырату, алюминий, мыс, цинк

РЕЗЮМЕ

Исследована возможность получения тонких интерметаллидных пленок, которые могут применяться в качестве терморегулирующих покрытий для космических аппаратов. Данные покрытия являются пассивной составной частью общей системы терморегулирования космических аппаратов, поэтому они должны обеспечивать высокую адгезию с материалом подложки и высокие функциональные характеристики, оптические в частности. Рассмотрены вопросы синтеза покрытий из устойчивых интерметаллидных фаз Al_4Cu_9 , Al_2Cu , Cu_5Zn_8 на алюминиевой и медной подложках путем послойного магнетронного распыления реагентов. Исследовались закономерности образования интерметаллидного покрытия в режимах «быстрого» и «медленного» напыления реагентов, различных толщин слоев напыляемых реагентов, температуры подложки, термообработки напыленного покрытия. Обнаружены режимы неполного и полного синтеза интерметаллидного покрытия. Получены снимки поперечного сечения покрытий, результаты микронзондового сканирования распределения реагентов по толщине напыленного покрытия, значения микротвердости, оптических коэффициентов поглощения и излучения, удельных электрических сопротивлений, адгезии к подложке. Получены прототипы интерметаллидных терморегулирующих покрытий классов «Солнечные отражатели» и «Солнечные поглотители». Результаты измерений оптических и прочностных характеристик показывают, что полученные интерметаллидные пленки могут применяться не только в качестве терморегулирующих покрытий для космической техники, но также и в общем машиностроении, благодаря высоким механическим свойствам.

Ключевые слова: солнце, аппарат, терморегуляция, покрытие, синтез, интерметаллиды, магнетрон, напыление, алюминий, медь, цинк.

Received 09.06.2016.